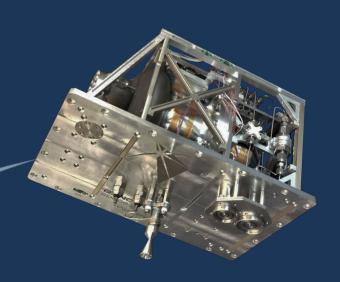


# HALCYON

**Integrated Non-Toxic Chemical Propulsion Systems** 

FIRST FLIGHT HERITAGE - 2021





### PRODUCT HIGHLIGHTS

Halcyon is a non-toxic ('green'), flight-proven high-thrust propulsion product line developed for 3U through ESPA satellite operations. Our systems are designed to remove common customer pain points by combining intelligent control electronics with a modular system architecture that utilizes readily available materials and propellants deliver highly configurable, cost-effective solutions with best-in-class lead times.

### **APPLICATIONS**



Insertion









**Transfer Avoidance** 

Keeping





**Momentum** Management

**RPO** and Servicing



Controlled **Deorbit** 

### MISSION-OPTIMIZED TECHNOLOGY

Based on mission-proven methods for catalytic combustion of High-Test Peroxide (HTP), our Halcyon product family offers a range of monopropellant or dual-mode (mono- and bipropellant operability) thrusters, propellant tank sizes, and system architectures, resulting in a well-equipped toolbox for mission-optimized mobility packages. The Halcyon product line designed for ease of integration utilizing standard interfaces that alleviate customer pain points and project schedule risk. From rapid maneuver operations to full missions, Halcyon is configured to enhance onorbit performance for any mission.

**Total Impulse:** 1000 - 500,000+ Ns

Thruster Quantity: 1, 4, 5, 8, Custom

**Optional:** ODPS™ on-demand pressurization

RAFTI™ refueling port

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# SYSTEM SPECIFICATIONS

### HALCYON & HALCYON AVANT

SYSTEM PICTURED ON PAGE 1



PARAMETER	CATALYTIC MONOPROPELLANT - HTP	DUAL-MODE BIPROPELLANT - HTP + FUEL INJECTION
THRUST	250 mN, 1 N, 5 N, 10 N	2 N, 10 N, 22 N
SPECIFIC IMPULSE	150 - 170 s	290 - 310 s
MAX FIRING TIME	10,000 s PER THRUSTER	10,000 s PER THRUSTER
MAX TOTAL THROUGHPUT (PER THRUSTER)	250mN: 1.56 kg/thruster 1N: 6.6 kg/thruster 5N: 32.5 kg/thruster 10N: 65.0 kg/thruster	2N: 6.8 kg/thruster 10N: 34.0 kg/thruster 22N: 74.8 kg/thruster
CATALYST BED PREHEAT - Optional	Optimal performance: 10 minutes before maneuver (enhances performance but not required)	
SYSTEM DIMENSIONS	Configuration Specific	
SYSTEM DRY MASS	Configuration Specific	
PROPELLANT MASS	.66 kg (per 1000 N·s Impulse)	.35 kg (per 1000 N·s Impulse)
MINIMUM IMPULSE	<10 mN·s	<35 mN·s
SYSTEM TEMP RANGE (SURVIVABLE)	-30°C to 60°C	
PROPELLANT TEMP (ON-ORBIT)	-5°C to 25°C	
AVERAGE POWER DRAW (IDLE/FIRING)	<4 W / <15 W	
PRESSURIZATION TIME	10-15 minutes for ODPS (one time event)  N/A if launching pre-pressurized (traditional pressurization)	
INPUT VOLTAGE RANGE	12V, 24V - 32V	24V - 32V
PHYSICAL LAYER INTERFACE	RS422/485	
PROTOCOL AND COMMAND INTERFACE	Modular - flexible to preferred customer architecture	
FIRST FLIGHT	June 2021	Sept 2022

## MORE MISSION. LESS COST.

Benchmark is a full lifecycle partner committed to supporting your mission from operational planning through asset decommission in LEO, GEO, and beyond. By combining our heritage propulsion products and advanced control systems with complementary products and services, Benchmark can deliver bundled in-space mobility solutions for 3U through ESPA and OTV spacecraft with significant cost, schedule, and capability benefits over alternative offerings.

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Contact us today to explore your mission!